



An investigation of flexible $\text{Cu}(\text{In},\text{Ga})\text{Se}_2$ solar cells under low intensity low temperature for potential application for outer planetary missions

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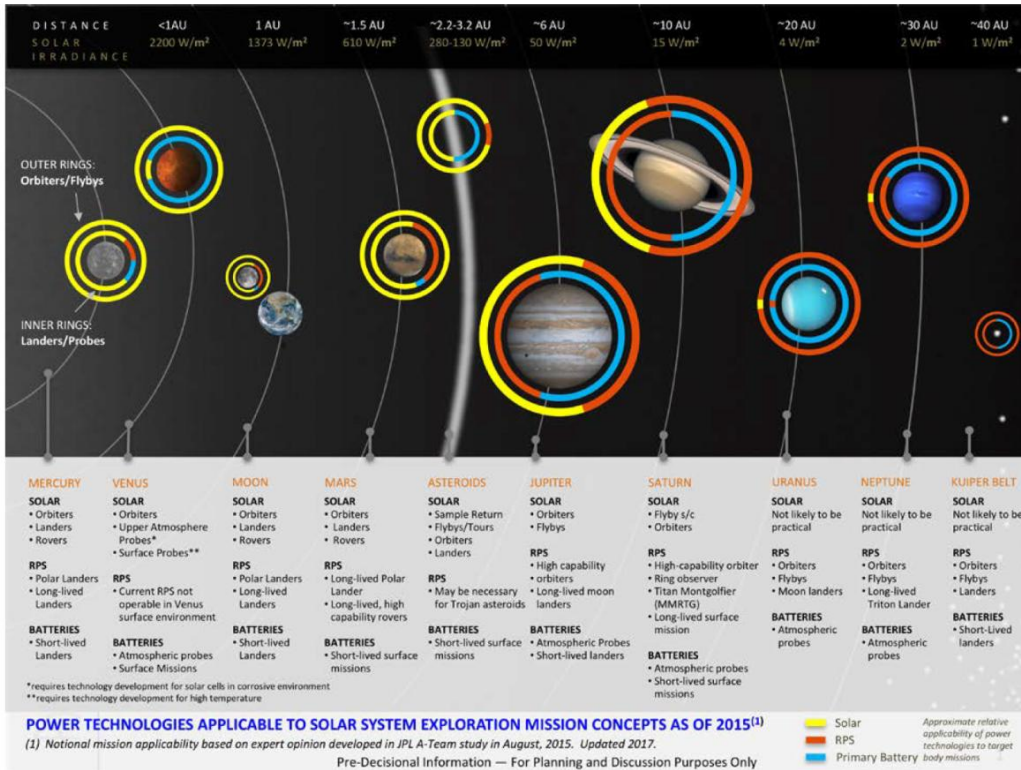
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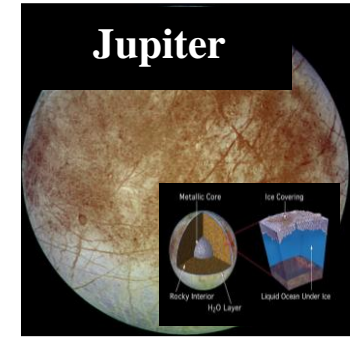
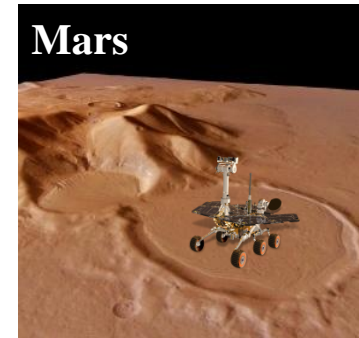




Photovoltaics for Next Generation Space Missions: Deep Space/Outer Planetary Missions



- NASA
- ESA
- JAXA
- Space X



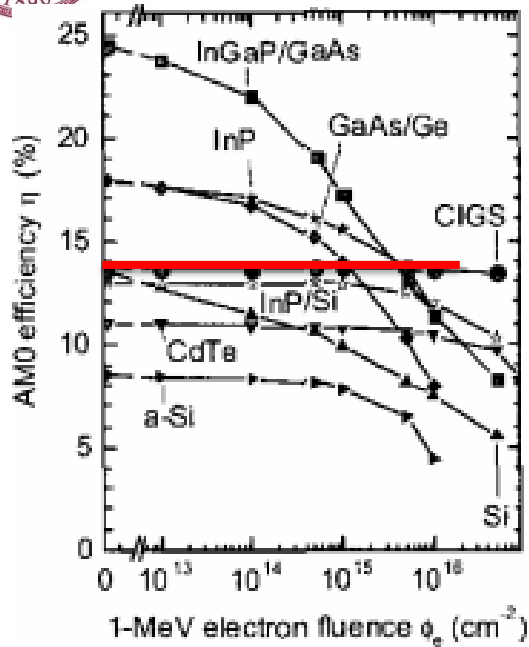
- Higher power requirements for outer planet exploration
 - Beyond power for most radioisotope thermoelectric generators (RTG)
- Outer planets have low temperature compared to Low Earth Orbit (LEO) and some missions, like those near Jupiter, will encounter intense radiation belts.
- Flexible radiation hard thin films solar cells may be competitive if packing ratio/specific power is high compared to multijunction
 - Particularly for low cost satellites (CubeSat and SmallSat, 6U and 24U)



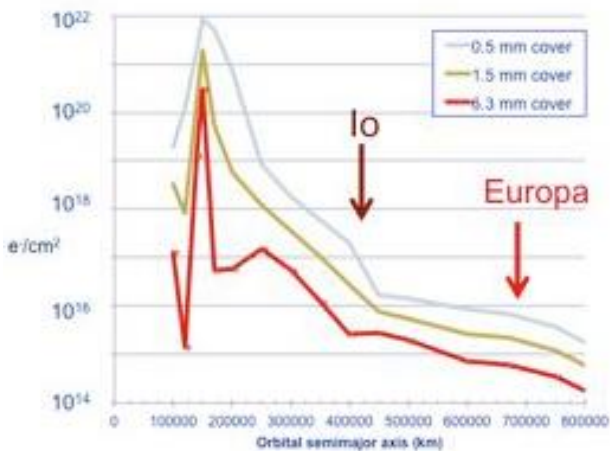
NASA/JPL/Caltech

G. A. Landis and J. Fincannon,
IEEE 42nd (PVSC),

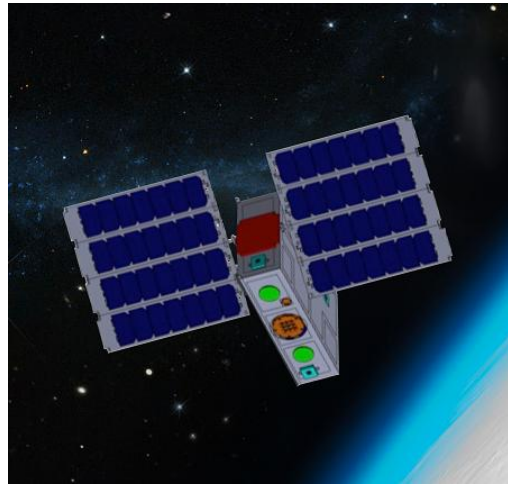
CIGS for Deep Space: a *unique application*



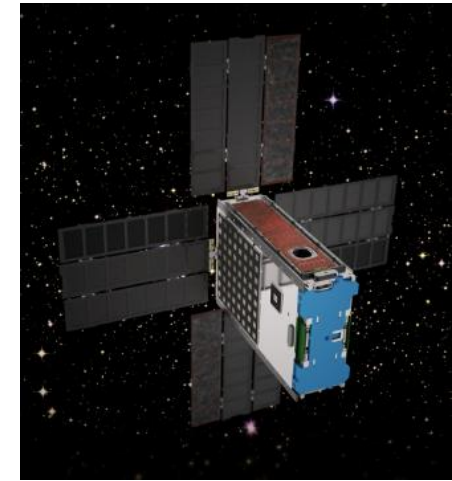
A. Jasenek *et al.*, Proc. WCPEC-3, 2003.



G. A. Landis and J. Fincannon, *IEEE 42nd (PVSC)*,



www.nasa.gov/mission_pages/smallsats



BioSentinel - www.nasa.gov

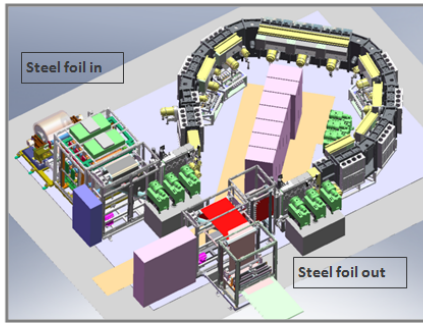
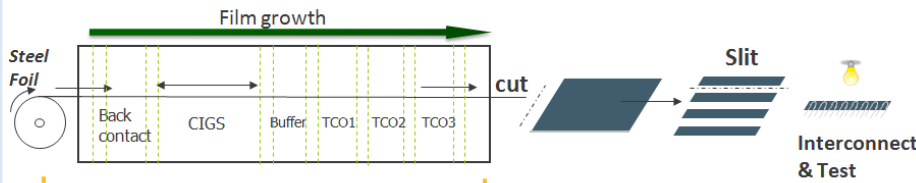
	I (AM0 suns)	I (W/m^2)	T_{eq} (K)
Saturn	0.011	14.82	100
Jupiter	0.037	50.26	135
Mars	0.431	586.2	263

Performance	NRA Goal
LILT BOL cell efficiency	35%
LILT EOL cell efficiency	28%
LILT EOL array specific power	8-10 W/kg
1AU BOL packaging density	60 kW/m^3
Stowed & launch capability	Yes
Operation in 100-300V range	Yes
Plasma operation 2eV, $1\text{e}8/\text{cm}^3$	Yes
LILT EOL minimum array power	5 kW

- Low cost, deployable technology
- (At least) equivalent payload
- Higher packing volume
- Radiation hard

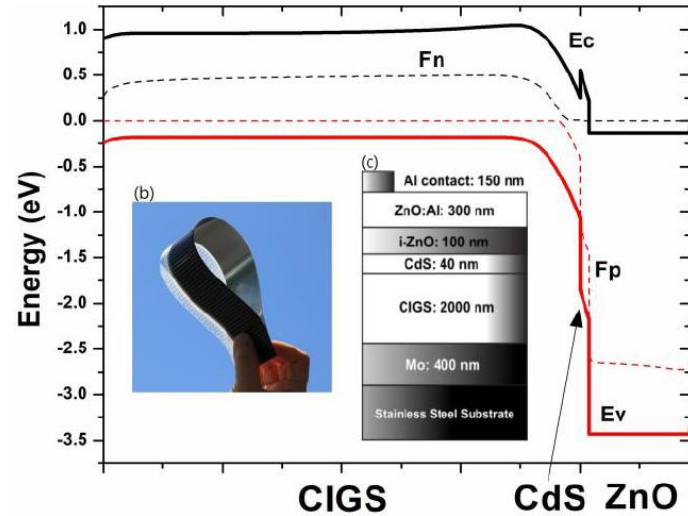
JPL (NASA) EESP Base Report 4/26/2017: “Solar Arrays for LILT and High Radiation Environments.”

MiaSolé Roll-to-Cell Process



- + Single interconnected vacuum chamber
- + High deposition rate sputtering process over 1 meter wide substrate
- + Cells finished with low resistance collection grid
- + Incoming material to IV data in ~60 minutes
- + Small factory footprint → fraction of a c-Si factory footprint

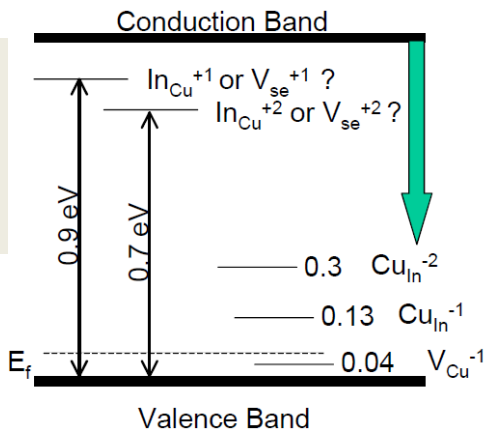
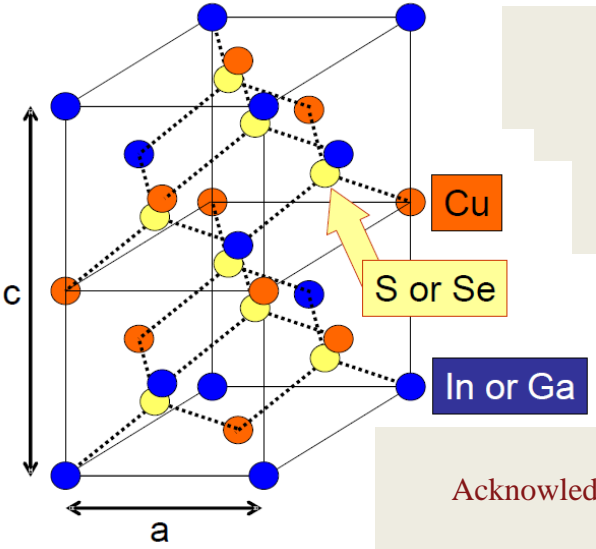
All PVD Deposition using Roll Coater



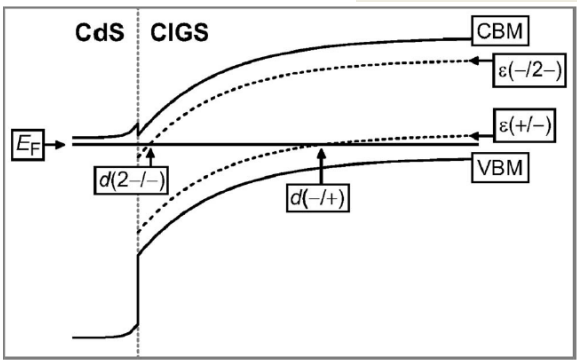
- Commercial grade CIGS with module efficiency of 17 % (20% - 2020)
- PVD Roll-to-Cell process on flexible steel
- Specifications: (for example - *FLEX-03WS*) 1.6 Kg/m²
- 1293 mm x 1010 mm = 190 W
- Payload/Specific Power (AM1.5G) ~ 90 W/kg



CIGS: Materials Properties: *Metastability*

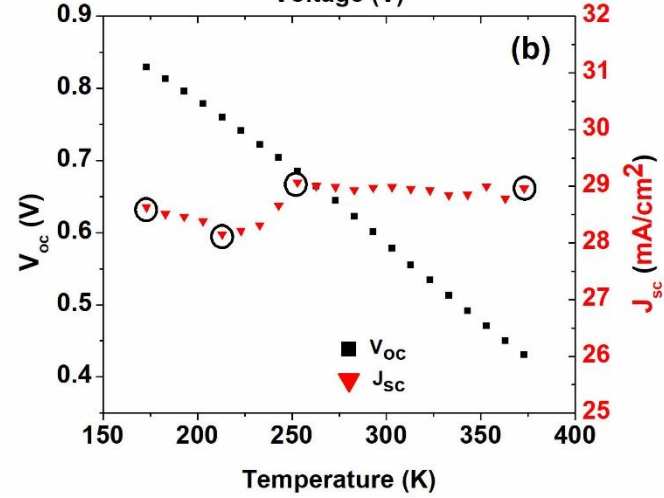
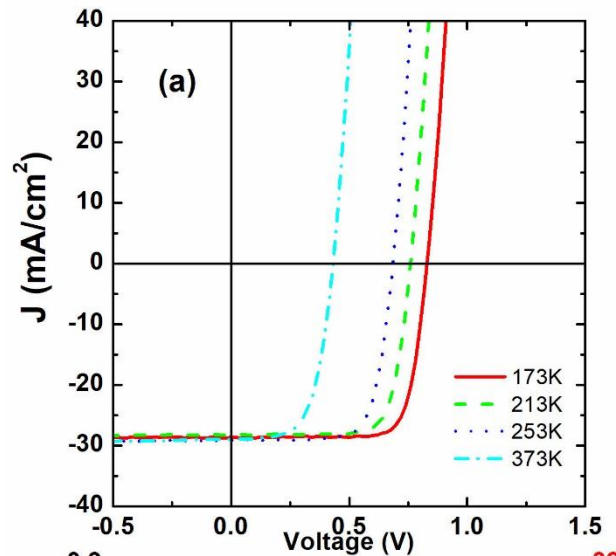
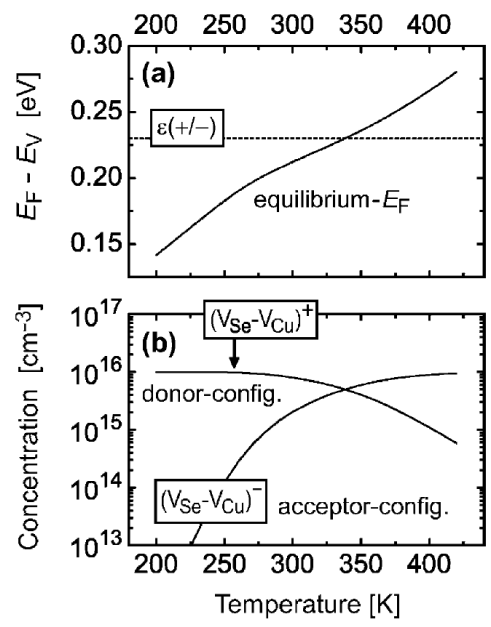


Acknowledgement - Angus Rockett (CSM)



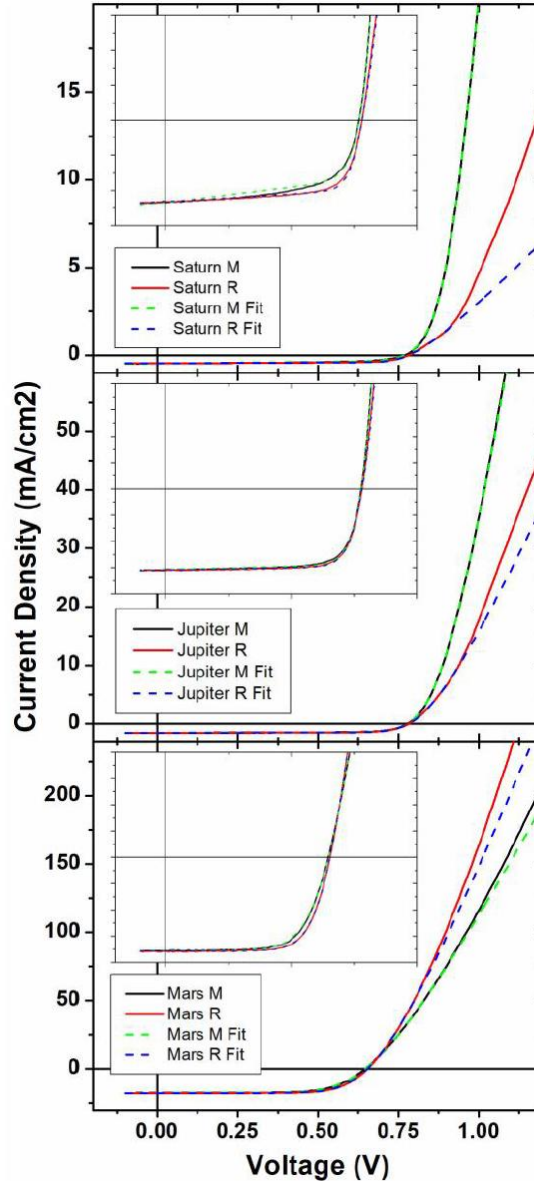
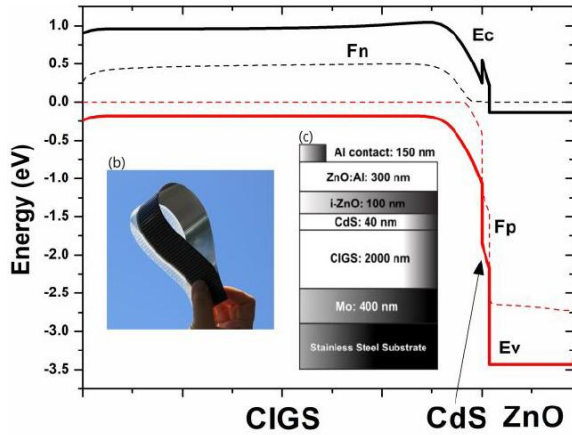
S. Lany and A. Zunger, *Journal of Applied Physics* **100** (11), 113725 (2006).

Reversible metastability under PV operating conditions



Brown *et al.* in preparation

Effects of Metastability: *LILT* Effect



$$I = I_0 \left(\exp \left(\frac{q(V - R_s I)}{nk_B T} \right) - 1 \right) + \frac{(V - R_s I)}{R_{sh}} - I_{ph}$$

Saturn: $T = 100 \text{ K}$; $I = 0.01 \text{ suns}$

- Loss of Fill factor in M-state
- Evidence of parasitic barrier

Jupiter: $T = 135 \text{ K}$; $I = 0.04 \text{ suns}$

- Loss of Fill factor in M-state (less than observed in Saturn)
- Higher thermal energy

Mars: $T = 263 \text{ K}$; $I = 0.4 \text{ suns}$

- Comparable fill factor (R and M)
- Reversal observed/ higher R_s in M-state
- Evidence of generation recombination losses in the bulk.

Brown *et al.* in preparation

- Relaxed – dark 330 K for 1 hour
- Metastable – light soaked at RT for 1 hour (AM0)

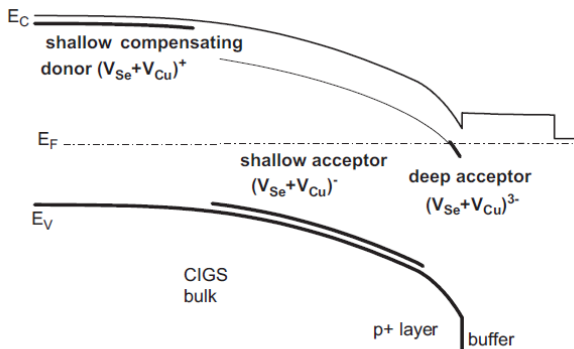
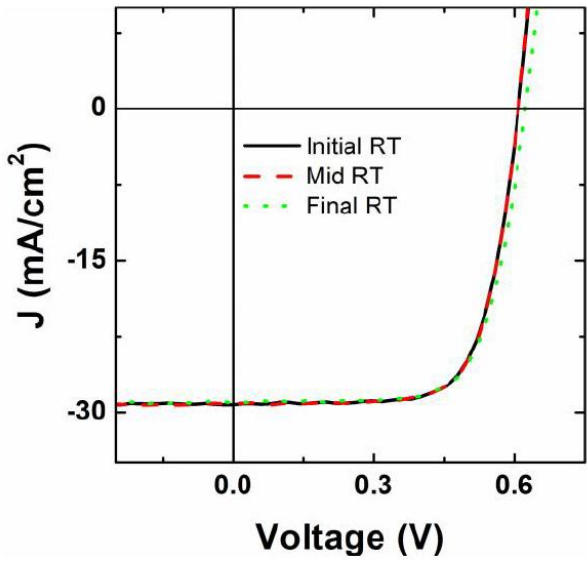


Fig. 1. Distribution of charged defects within the junction according to the Lany-Zunger model of $V_{Se}-V_{Cu}$ divacancy [9,10].

M. Igalson *et al.*, *Solar Energy Materials and Solar Cells* **93** (8), 1290 (2009).



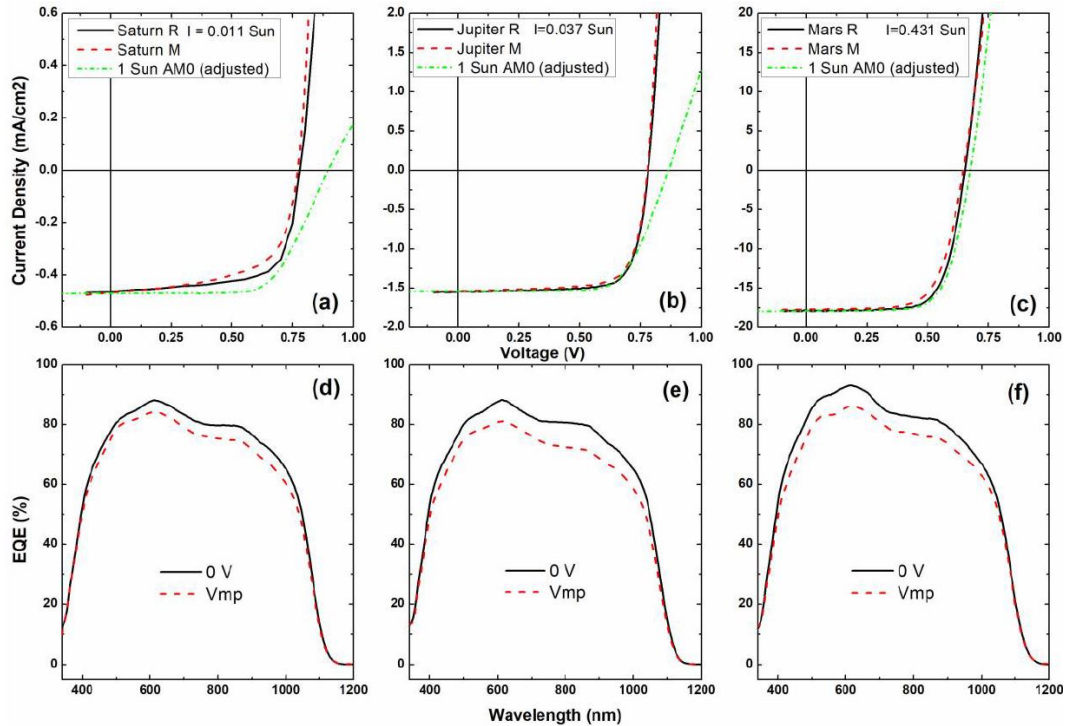
Thermal Cycling and LILT Analysis



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- Initial – AM1.5G 300 K
- Mid RT – after 12 hour at -100 °C
- Final RT – after 12 hour at 100 °C

No significant degradation – some improvement after high temperatures!



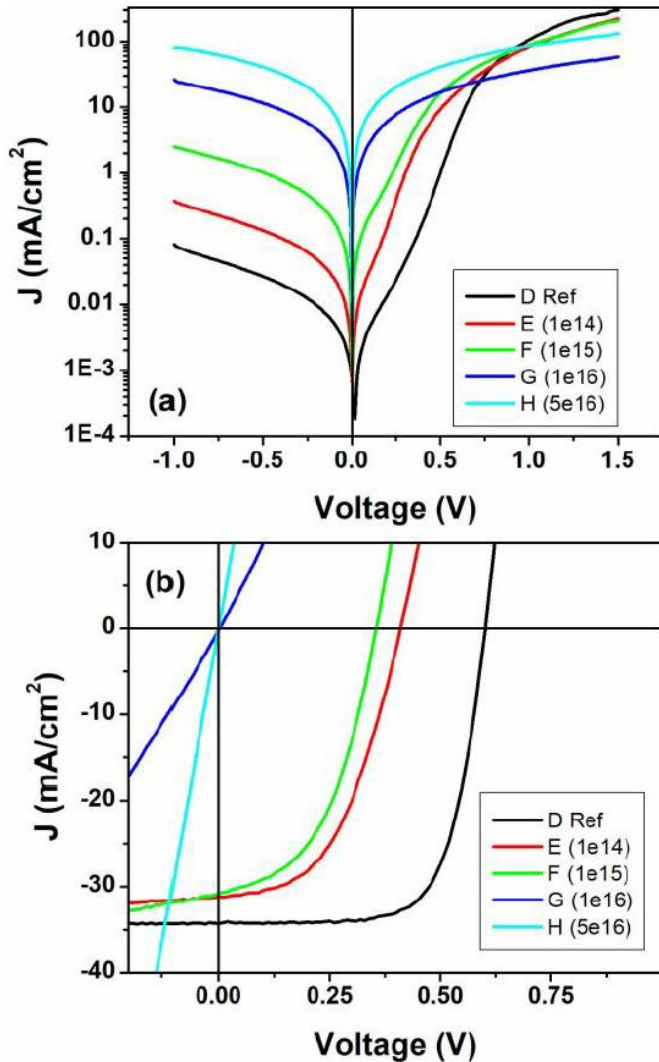
$$V_{oc} = \frac{nkT}{q} \ln\left(\frac{I_l}{I_0} + 1\right)$$

- Solar Cells measured at conditions equivalent to Saturn, Jupiter, and Mars
- Loss of voltage in lower LILT conditions compared with AM0 – *metastable defects/impurities*
- Evidence of photosensitive barrier at lower temperatures
- EQE suggest losses are Voltage related

LILT EOL array specific power	8-10 W/kg
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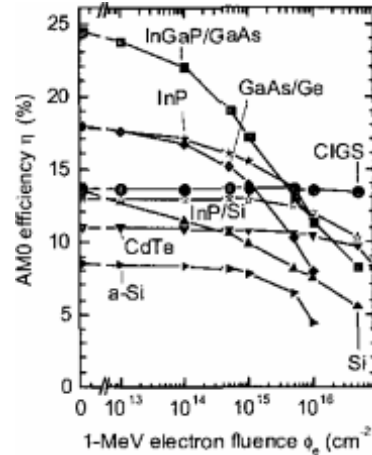
- Initial AM0 PCE ~ 13-14%
- LILT AM0 PCE ~ 19%
- LILT specific power at 19% PCE – 6.2 W/kg

Effects of proton irradiation



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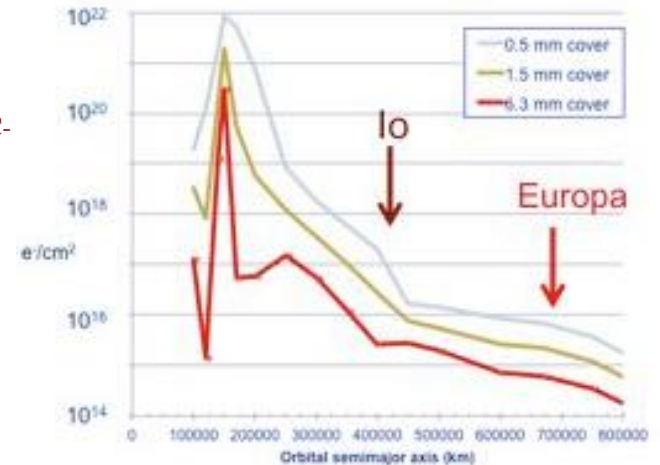
- Solar cells exposed to 1MeV proton irradiation/fluence from 1×10^{14} protons/cm² to 1×10^{16} protons/cm² – no encapsulation
- Rapid degradation evident....
- Significantly higher than typically used!



A. Jasenek *et al.*, Proc. WCPEC-3, 2003.

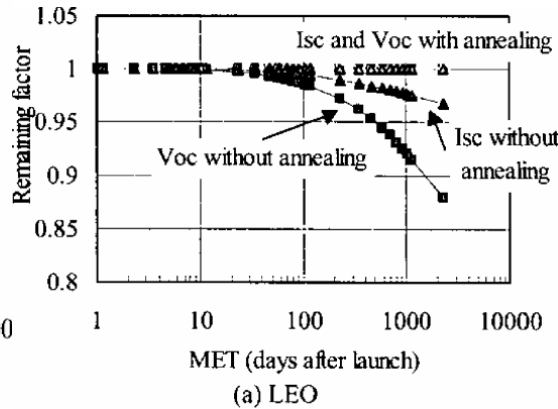
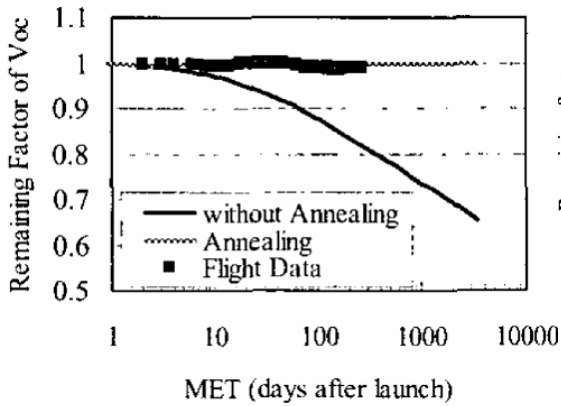
Fluence (e/cm ²)	Efficiency 5AU -125°C	Efficiency 1.58AU -125°C	Efficiency 1AU 28°C
0e00 (Ctrl)	37.6% ± 0.7%	39.0% ± 0.5%	33.0% ± 0.4%
1e15 (Rad)	35.0% ± 0.6%	36.2% ± 0.7%	27.0% ± 0.4%
4e15 (Rad)	27.9% ± 0.4%	29.6% ± 0.7%	20.8% ± 0.2%

JPL (NASA) EESP Base Report 4/26/2017: “Solar Arrays for LILT and High Radiation Environments.”

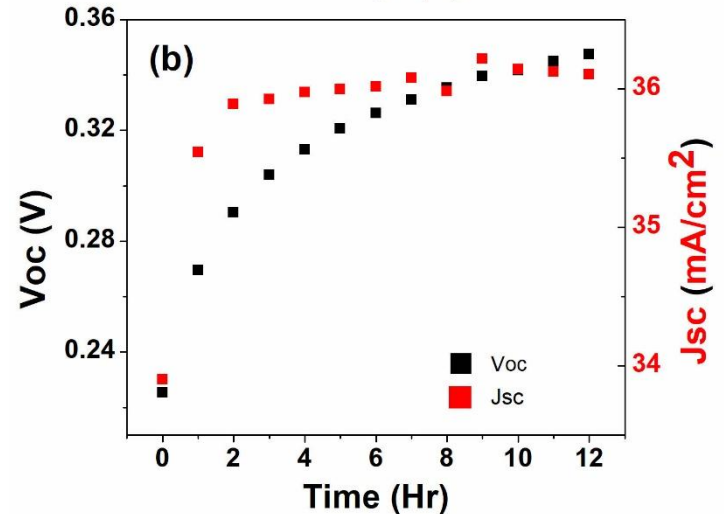
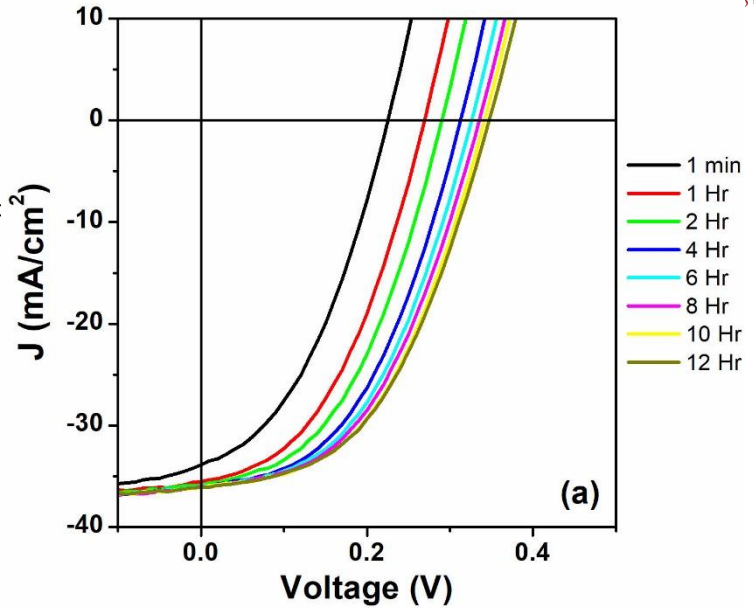




Self-Healing behavior of proton irradiated samples



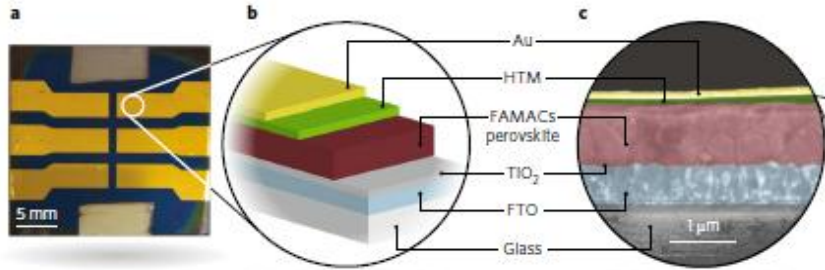
Kawakita *et al.* WCPEC 2003



Brown *et al.* in preparation

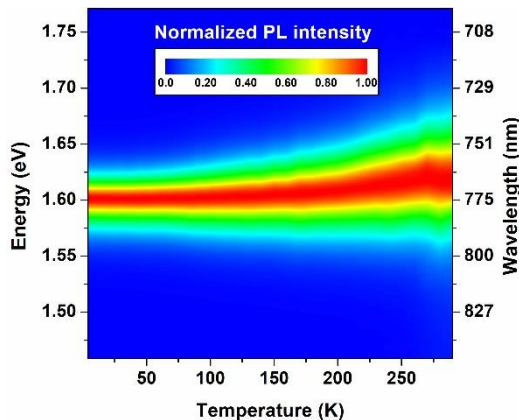
- MDS-1 satellite showed CIGS cells healed from radiation by annealing
- Cells exposed to heat (100C) under AM0 illumination
- Upon heating strong evidence of “self-healing”
- Further studies underway

LILT studies of promising thin film solar

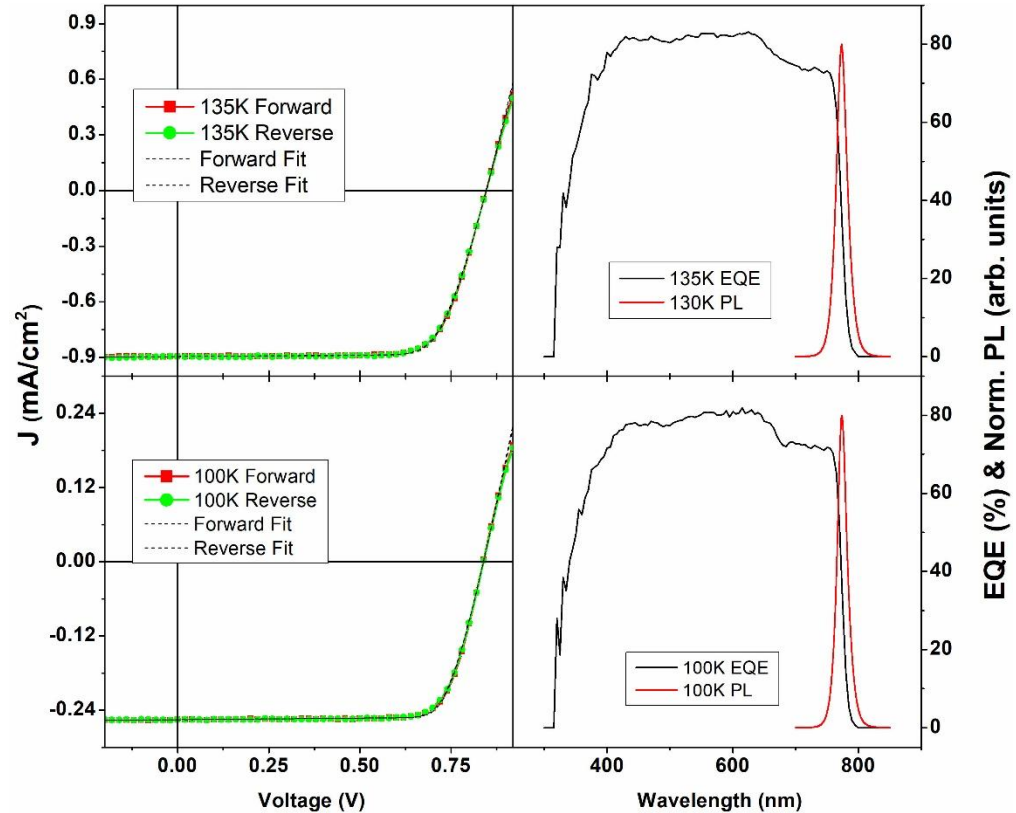


J. A. Christians *et al.*, Nature Energy 3 (1), 68 (2018).

- LILT studies of FAMACs perovskite
- High stability



- Show some promise for LILT space applications



Brown *et al.* in preparation



Credit: Giles Eperon



Summary and Acknowledgements

- For future trips to deep space locations, technologies unique to the rigors of those environments need to be developed
- CIGS appear to have unique potential for deep space CubeSat and SmallSat applications

- K. Hossain
- T. D. Golding



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